



# OEM-IMU-HG4930

Small, affordable MEMS IMU combines with SPAN GNSS+INS technology from Hexagon | NovAtel to provide 3D position, velocity and attitude

## World-leading GNSS+INS technology

SPAN GNSS+INS technology brings together two different but complementary technologies: Global Navigation Satellite System (GNSS) positioning and Inertial Navigation Systems (INS). The absolute accuracy of GNSS positioning and the stability of Inertial Measurement Unit (IMU) gyro and accelerometer measurements combine to provide an exceptional 3D navigation solution that is stable and continuously available, even through periods when satellite signals are blocked.

## Sophisticated, tactical grade MEMS performance

The OEM-IMU-HG4930 is a high performing Micro Electromechanical Systems (MEMs) IMU. Economical, robust and small, the low power OEM-IMU-HG4930 provides tactical grade performance for unmanned vehicles and commercial and/or military guidance applications. When integrated with SPAN GNSS+INS technology, this IMU is ideal for airborne, marine and ground applications that require accurate 3D position, velocity and attitude data in a compact package.

## **OEM-IMU-HG4930** advantages

The OEM-IMU-HG4930 is comprised entirely of commercial components, simplifying export processes for this IMU.

## Improved accuracy

Receivers from NovAtel provide your choice of accuracy and performance, from decimetre to RTK-level positioning. For the most demanding applications, Waypoint Inertial Explorer post-processing software offers the highest level of accuracy.



## Benefits

- High performance IMU
- Optimal for aerial, hydrographic survey and industrial applications
- Easy integration with NovAtel's SPAN capable GNSS+INS receivers
- Commercially exportable
- Low 5 VDC power input

## Features

- MEMS gyros and accelerometers
- Small size, rugged and light weight
- IMU data rate: 100 Hz
- Direct UART interface to OEM7
  receivers
- SPAN GNSS+INS capability with configurable application profiles

#### OEM-IMU-HG4930 Product Sheet

#### SPAN System Performance<sup>1</sup>

#### Horizontal Position Accuracy (RMS)

Single Point L1	1.5 m	Techno
Single Point L1/L2	1.2 m	Input ra
SBAS <sup>2</sup>	60 cm	Full P
DGPS	40 cm	Full O
TerraStar-L <sup>3,4</sup>	40 cm	
TerraStar-C PRO <sup>3, 4</sup>	2.5 cm	Accele
TerraStar-X <sup>3, 4</sup>	2 cm	Techno
RTK	1cm+1ppm	Range
Data Rate		
IMU Raw Data Rate INS Solution	100 Hz or 400 Hz Up to 200 Hz	
Time Accuracy <sup>4</sup>	20 ns RMS	

515 m/s

Max Velocity<sup>5</sup>

### IMU Performance<sup>6</sup>

#### **Gyroscope Performance**

echnology	MEMS	
nput rate		
Full Performance Range		±325°/sec
Full Operating Range		±400°/sec
ccelerometer P	erformance	

Accelerometer Performance Technology MEMS

Technology MEMS Range ±20 g Physical and Electrical<sup>6</sup>

**Dimensions** 64.8 mm dia max × 35.7 mm h max

Weight 200 g

#### Power

Power consumption <3 W (typical) Input voltage +5 VDC

#### **Environmental**<sup>6</sup>

Temperature

Operating Storage -40°C to +71°C -40°C to +80°C

Random Vibe

MIL-STD-810G(Ch1), Method 514.7 (2.0g)

#### Performance During GNSS Outages7

Outage Duration	Positioning Mode	Position Accuracy (M) RMS		Velocity Accuracy (M/S) RMS		Attitude Accuracy (Degrees) RMS		
		Horizontal	Vertical	Horizontal	Vertical	Roll	Pitch	Heading
0 s	RTK <sup>8</sup>	0.02	0.03	0.015	0.010	0.010	0.010	0.030
	PPP	0.06	0.15					
	SP	1.00	0.60					
	Post-Processed <sup>9</sup>	0.01	0.02	0.015	0.010	0.003	0.003	0.010
10 s	RTK <sup>8</sup>	0.12	0.08	0.035	0.020	0.018	0.018	0.040
	PPP	0.16	0.20					
	SP	1.10	0.65					
	Post-Processed <sup>9</sup>	0.01	0.02	0.015	0.010	0.003	0.003	0.010
60 s	RTK <sup>8</sup>	3.82	0.73	0.165	0.030	0.030	0.030	0.055
	PPP	3.86	0.85					
	SP	4.80	1.30					
	Post-Processed <sup>9</sup>	0.11	0.05	0.017	0.010	0.004	0.004	0.014

1. Typical values. Performance specifications subject to GNSS system characteristics, Signal-In-Space (SIS) operational degradation, ionospheric and tropospheric conditions, satellite geometry, baseline length, multipath effects and the presence of intentional or unintentional interference sources. 2. GPS-only. 3. Requires subscription to TerraStar data service. Subscriptions available from NovAtel. 4. Time accuracy does not include biases due to AF or antenna delay. 5. Export licensing restricts operation to a maximum of SIS metres/second. 6. Supplied by IMU damafcaturer. 7. Outrage statistics were calculated by taking the RMS of the maximum errors over a minimum of 30 complete GNSS autages. Each autage was followed by 120 seconds of full GNSS availability before the next outage was applied. High accuracy GPS updates (fixed ambiguities) were available immediately before and after each outage. The survey data used to generate these statistics is ground vehicle data callected with frequent changes in azimuth (i.e., as normally observed in ground vehicle environments). 8. Topm should be added to all values to account for additional error due to baseline length. 9. Post-processing results using inderial Explorer. The survey data used to generate these statistics had frequent changes in azimuth.

## Contact Hexagon | NovAtel

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